2017-2018 MM2MS3 Exam Solutions

Vertical equilibrium of the beam:

1.

Taking moments about position C:

Substituting values of P_o and M_o gives:

Rearranging (1) for
$$R_c$$
 and substituting values for R_A and P_o gives:

$$R_{C} = 3,500 \text{ N}$$

[1 mark]

Taking the origin at the left-hand end of the beam, sectioning after the last discontinuity and drawing a free body diagram of the left-hand side of the section:





 $R_A + R_C = P_o$

 $P_o \times 1 = R_A \times 2 + M_o$

 $\therefore R_A = \frac{P_o - M_o}{2}$

 $R_A = 1,500 \text{ N}$

(1)









[3 marks]

Taking moments about the section position:

$$R_A \langle x - 1 \rangle + M_o \langle x - 2 \rangle^0 = M + P_o \langle x - 2 \rangle$$
$$\therefore M = R_A \langle x - 1 \rangle + M_o \langle x - 2 \rangle^0 - P_o \langle x - 2 \rangle$$

[2 marks]

Substituting this into the main deflections of beams equation ($EI \frac{d^2y}{dx^2} = M$):

$$EI\frac{d^2y}{dx^2} = R_A \langle x - 1 \rangle + M_o \langle x - 2 \rangle^0 - P_o \langle x - 2 \rangle$$

[2 marks]

Integrating with respect to *x*:

$$EI\frac{dy}{dx} = \frac{R_A \langle x - 1 \rangle^2}{2} + M_o \langle x - 2 \rangle - \frac{P_o \langle x - 2 \rangle^2}{2} + A$$
(2)

[2 marks]

Integrating with respect to x again:

$$EIy = \frac{R_A \langle x - 1 \rangle^3}{6} + \frac{M_o \langle x - 2 \rangle^2}{2} - \frac{P_o \langle x - 2 \rangle^3}{6} + Ax + B$$
(3)

[2 marks]

Boundary conditions:

(BC1) At x = 1, y = 0, therefore from (3):

$$A + B = 0 \tag{4}$$

[2 marks]

(BC2) At x = 3, y = 0, therefore from (3):

$$0 = \frac{R_A \times 2^3}{6} + \frac{M_o \times 1^2}{2} - \frac{P_o \times 1^3}{6} + 3A + B$$
$$\therefore A = \frac{P_o - 3M_o - 8R_A}{12}$$

Substituting values of P_o , R_A and M_o into this gives:

$$A = -1,083.33$$



Therefore, from (4):

$$B = 1,083.33$$

[2 marks]

From (2), at x = 2 (point B):

 $\frac{dy}{dx} = \frac{1}{EI} \left(\frac{R_A \langle x - 1 \rangle^2}{2} + M_o \langle x - 2 \rangle - \frac{P \langle x - 2 \rangle^2}{2} + A \right)$

Substituting values of E, I, R_A , M_o , P_o and A into this gives:

 $\frac{dy}{dx} = -0.000833 \text{ rads}$

[2 marks]

From (3), at x = 2 (point B):

$$y = \frac{1}{EI} \left(\frac{R_A (x-1)^3}{6} + \frac{M_o (x-2)^2}{2} - \frac{P_o (x-2)^3}{6} + Ax + B \right)$$

Substituting values of E, I, R_A , M_o , P_o , A and B into this gives:

y = 0.001042 m = 1.042 mm

(i.e. upward deflection)



2.

(a)

Position of Centroid, C



Total area,

$$A = (50 \times 15)_a + (15 \times 45)_b + (50 \times 15)_c = 2,175 \text{ mm}^2$$

Taking moments about AA:

$$\bar{y} = \frac{(50 \times 15 \times 7.5)_a + (15 \times 45 \times 37.5)_b + (50 \times 15 \times 67.5)_c}{2,175}$$
$$\therefore \bar{y} = 37.5 \text{ mm}$$

[3 marks]

Similarly, taking moments about BB:

$$\bar{x} = \frac{(15 \times 50 \times 25)_a + (45 \times 15 \times 42.5)_b + (15 \times 50 \times 60)_c}{2,175}$$

 $\therefore \overline{x} = 42.5 \text{ mm}$

[3 marks]



(b)

Principal 2nd Moments of Area

Using the Parallel Axis Theorem,

$$I_{x'} = (I_x + Ab^2)_a + (I_x + Ab^2)_b + (I_x + Ab^2)_c$$
$$= \left(\frac{50 \times 15^3}{12} + 50 \times 15 \times 30^2\right) + \left(\frac{15 \times 45^3}{12} + 15 \times 45 \times 0^2\right) + \left(\frac{50 \times 15^3}{12} + 50 \times 15 \times -30^2\right)$$
$$= 1,492,031.25 \text{ mm}^4$$

[2 marks]

and,

$$I_{y'} = (I_y + Aa^2)_a + (I_y + Aa^2)_b + (I_y + Aa^2)_c$$
$$= \left(\frac{15 \times 50^3}{12} + 15 \times 50 \times -17.5^2\right) + \left(\frac{45 \times 15^3}{12} + 45 \times 15 \times 0^2\right) + \left(\frac{15 \times 50^3}{12} + 15 \times 50 \times 17.5^2\right)$$
$$= 784,531.25 \text{ mm}^4$$

[2 marks]

Also,

$$I_{x'y'} = (I_{xy} + Aab)_a + (I_{xy} + Aab)_b + (I_{xy} + Aab)_c$$
$$= (0 + 50 \times 15 \times -17.5 \times 30) + (0 + 15 \times 45 \times 0 \times 0) + (0 + 50 \times 15 \times 17.5 \times -30)$$
$$= -787,500 \text{ mm}^4$$



Mohr's Circle:



[3 marks]

Therefore, the Principal 2nd Moments of Area are:

$$I_P = C + R = 1,138,281.25 + 863,304.88$$

 $\therefore I_P = 2,001,586.13 \text{ mm}^4$

[2 marks]

and,

$$I_Q = C - R = 1,138,281.25 - 863,304.88$$

 $\therefore I_Q = 274,976.37 \text{ mm}^4$



(c)

Orientation of the Principal Axes with respect to the x-y co-ordinate system

From the Mohr's circle above:

$$sin2\theta = \frac{I_{xy}}{R} = \frac{-787,500}{863,304.88}$$

 $\therefore \theta = -32.91^{\circ}$

[3 marks]

Therefore, **the Principal Axes are at -32.91° (anti-clockwise) from the** *x***-***y* **axes**, as shown on the diagram below.



[3 marks]



3.

(-)

(a)

$$\sigma_r = A - \frac{B}{r^2} - \frac{3+\nu}{8}\rho\omega^2 r^2$$
$$\sigma_\theta = A + \frac{B}{r^2} - \frac{1+3\nu}{8}\rho\omega^2 r^2$$

[2 marks]

at r = 0.05 (ID), $\sigma_r = 0$ therefore:

$$0 = A - \frac{B}{0.05^2} - \frac{3 + 0.3}{8}7900 \times 523.6^2 \times 0.05^2$$

$$\therefore 0 = A - 400B - 223.4 \times 10^4$$
(1)

or,

[3 marks]

at r = 0.5 (OD), $\sigma_r = 0$ therefore:

$$0 = A - \frac{B}{0.5^2} - \frac{3 + 0.3}{8}7900 \times 523.6^2 \times 0.5^2$$

$$\therefore 0 = A - 4B - 223.4 \times 10^6$$
(2)

or,

 $A-4B-814.6\omega^2$

 $A - 400B - 8.15\omega^2$

[3 marks]

substituting (2) from (1):

 $0 = -396B + 221.16 \times 10^6$

[1 mark]

therefore:

 $B = 558500 \text{ (or } 2.04\omega^2\text{)}$

and from (1),

 $A = 225.6 \times 10^{6} \text{ (or } 822.7\omega^2)$



r	sigma_r (MPa)	sigma_theta (MPa)
5.00E-02	0.00E+00	4.48E+02
0.15	1.81E+02	2.39E+02
0.3	1.39E+02	1.85E+02
0.5	0.00E+00	9.92E+01

[2 marks]



[2 marks]

(b)

Max sigma theta is at bore (r = 0.05), therefore:

$$\sigma_{\theta_{max}} = 822.7\omega^2 + \frac{2.04\omega^2}{r^2} - \frac{1+3\times0.3}{8}7900\omega^2 0.05^2$$
$$\sigma_{\theta_{max}} = \omega^2 \left(822.7 + \frac{2.04}{0.05^2} - 1185\times0.05^2\right) = 1635.7\omega^2$$

[4 marks]

Hoop stress is limited to 240 MPa, therefore at the ID,

$$240 \times 10^{6} = 1635.7\omega^{2}$$

 $\omega^{2} = 146723$



$$383 \text{ rad/s} = \frac{\omega}{2\pi} \times 60 = 3657 \text{ rpm}$$

[6 marks]



4.

(a)

The shear stress distribution in the web is given by:

$$\tau_{flange} = \frac{SA\bar{y}}{Iz} = \frac{S(at)d}{2It} = \frac{Sda}{2I}$$

where,

$$I = \frac{bd^3 - (d - 2t)^3(b - t)}{12}$$

therefore,

$$\tau_{flange} = \frac{6Sda}{bd^3 - (d - 2t)^3(b - t)}$$

14 marksi	[4	marks]
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at free surface,

 $\tau_{flange0} = 0$

[2 marks]

$$\therefore \tau_{flange28} = \frac{6 \times 5000 \times 40 \times 28}{30 \times 40^3 - (40 - 2 \times 2)^3 (30 - 2)} = 54.8 \text{ MPa}$$

[2 marks]

(b)

$$\tau_{web} = \frac{SA\bar{y}}{Iz} = \frac{S}{It} \left[\frac{btd}{2} + \left(\frac{d}{2} - y\right) t \left(\frac{d}{2} + y\right) \frac{1}{2} \right] = \frac{S}{2I} \left(bd + \left(\frac{d}{2}\right)^2 - y^2 \right)$$
$$\therefore \tau_{web} = \frac{6S}{bd^3 - (d - 2t)^3(b - t)} \left(bd + \left(\frac{d}{2}\right)^2 - y^2 \right)$$

[4 marks]

$$\tau_{web0} = \frac{6 \times 5000}{30 \times 40^3 - (40 - 2 \times 2)^3 (30 - 2)} \times \left((30 \times 40) + \left(\frac{40}{2}\right)^2 - 0^2 \right) = \mathbf{78.2} \text{ MPa}$$



$$\tau_{web18} = \frac{6 \times 5000}{30 \times 40^3 - (40 - 2 \times 2)^3 (30 - 2)} \times \left((30 \times 40) + \left(\frac{40}{2}\right)^2 - 19^2 \right) = 62.4 \text{ MPa}$$
[2 marks]

(c)

The force given by the shear stress in the flange (S_1) is;

$$S_1 = \int_0^b \tau t da = \int_0^b \frac{Sda}{2I} t da = \frac{Sdtb^2}{4I}$$

[3 marks]

If we take moments about O in the web (where e is the distance along the N.A. from O):

$$Se = 2S_1 \frac{d}{2}$$

[3 marks]

Therefore:

$$e = \frac{S_1 d}{S} = \frac{d^2 t b^2}{4l} = \frac{3 \times (40^2 \times 2 \times 30^2)}{(30 \times 40^3 - (40 - 2 \times 2)^3 (30 - 2))} = 14.1$$
mm

[3 marks]



5.

(a)

Below is a diagrammatic representation of a pinned-pinned strut.



Sectioning this beam in order to determine the bending moment:



[1 mark]

Taking moments about the section position, X-X:

$$M = Py \tag{1}$$

 2^{nd} order differential equation for a beam under bending:

$$EI\frac{d^2y}{dx^2} = M$$

Substituting (1) into this:

Let $y = A_0 e^{\alpha x}$:

and:

We get:

where A_0 , A and B are constants.

Boundary conditions:

(BC1) At x = 0, y = 0, therefore from (2):

(BC2) At x = L, y = 0, therefore from (2):

Since $A \neq 0$ for non-trivial solution:

$$\sqrt{\frac{P}{EI}}L = n\pi$$

14



[1 mark]

[1 mark]

[1 mark]

[± mai

B = 0

 $Asin\left(\sqrt{\frac{P}{EI}}L\right) = 0$

[1 mark]

,

(2)

 $\therefore EI\alpha^2 + P = 0$

 $\alpha = \pm \sqrt{\frac{P}{EI}}i$

 $y = Asin\left(\sqrt{\frac{P}{EI}}x\right) + Bcos\left(\sqrt{\frac{P}{EI}}x\right)$



vel.1

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 $\therefore P = \frac{n^2 \pi^2 EI}{L^2}$

where n = 1, 2, ...

(b)

Substituting the buckling load (from (3)) into equation (4):

 $\sigma = \frac{n^2 \pi^2 EI}{L^2 A} = \frac{n^2 \pi^2 EAk^2}{L^2 A} = \frac{\pi^2 E}{\left(\frac{L}{k}\right)^2}$

Therefore, when $\sigma = \sigma_{\gamma}$:

The following figure therefore displays the transition between yielding and buckling in terms of slenderness ratio.

15

[2 marks]







[2 marks]

(3)

(4)

 $\frac{L}{k} = \sqrt{\frac{\pi^2 E}{\sigma_y}}$

 $\sigma = \frac{P}{A}$

[2 marks]

[2 marks]

[4 marks]



(c)

In the fixed-fixed case,

$$P = \frac{4\pi^2 EI}{L^2}$$

Substituting this into equation (3):

$$\sigma = \frac{4\pi^{2} EI}{L^{2} A} = \frac{4\pi^{2} EAk^{2}}{L^{2} A} = \frac{4\pi^{2} E}{\left(\frac{L}{k}\right)^{2}}$$

Therefore, when $\sigma = \sigma_y$:

$$\frac{L}{k} = \sqrt{\frac{4\pi^2 E}{\sigma_y}} = 2\pi \sqrt{\frac{200,000}{250}} = 177.72$$
(5)

where k is calculated by:

$$I = \frac{bd^3}{12} = \frac{t^4}{12} = t^2k^2$$

where b = d = t and $I = Ak^2 = bdk = t^2k^2$.

$$\therefore k = 12.99 \text{ mm}$$

Therefore from (5):

L = 2,308.58 mm = 2.309 m



6.

Second Moment of Area, *I*, calculation:



[2 marks]

Adding dummy load, Q, and labelling the structure:



[2 marks]

Section AB (bending only)

Free Body Diagram:







Taking moments about X-X:

$$M_{AB} = Pa + Qb = Px\cos\theta + Qx\sin\theta$$

[1 mark]

Substituting this into the equation for Strain Energy in a beam under bending gives,

$$\begin{aligned} U_{AB} &= \int \frac{M_{BC}^2}{2EI} ds = \int_0^L \frac{(Px\cos\theta + Qx\sin\theta)^2}{2EI} dx = \frac{1}{2EI} \int_0^L (P^2x^2\cos^2\theta + Q^2x^2\sin^2\theta + PQx^2\cos\theta\sin\theta) dx \\ &= \frac{(P^2\cos^2\theta + Q^2\sin^2\theta + PQ\cos\theta\sin\theta)}{2EI} \int_0^L x^2 dx = \frac{(P^2\cos^2\theta + Q^2\sin^2\theta + PQ\cos\theta\sin\theta)}{2EI} \left[\frac{x^3}{3}\right]_0^L \\ &\therefore U_{AB} = \frac{L^3}{6EI} (P^2\cos^2\theta + Q^2\sin^2\theta + PQ\cos\theta\sin\theta) \end{aligned}$$

[2 marks]

Section BC (bending only)

Free Body Diagram:



[2 marks]

Taking moments about Y-Y:

$$M_{BC} + Qd = Pc$$

$$\therefore M_{BC} = Pc - Qd = P(x - Lcos\theta) - QLsin\theta$$

[1 mark]



Substituting this into the equation for Strain Energy in a beam under bending gives,

$$\therefore U_{BC} = \int \frac{M_{BC}^2}{2EI} ds = \int_0^L \frac{(P(x - L\cos\theta) - QL\sin\theta)^2}{2EI} dx$$

$$= \frac{1}{2EI} \int_0^L (P^2 x^2 - 2P^2 Lx\cos\theta + P^2 L^2 \cos^2\theta - 2PQLx\sin\theta + Q^2 L^2 \sin^2\theta + 2PQL^2 \cos\theta\sin\theta) dx$$

$$= \frac{1}{2EI} \left[\frac{P^2 x^3}{3} - P^2 Lx^2 \cos\theta + P^2 L^2 x\cos^2\theta - PQLx^2 \sin\theta + Q^2 L^2 x\sin^2\theta + 2PQL^2 x\cos\theta\sin\theta \right]_0^L$$

$$\therefore U_{BC} = \frac{L^3}{2EI} \left(\frac{P^2}{3} - P^2 \cos\theta + P^2 \cos^2\theta - PQ\sin\theta + Q^2 \sin^2\theta + 2PQ\cos\theta\sin\theta \right)$$

[2 marks]

Total Strain Energy

$$U = U_{AB} + U_{BC}$$

$$\therefore U = \frac{L^3}{6EI} (P^2 \cos^2\theta + Q^2 \sin^2\theta + PQ \cos\theta \sin\theta)$$

$$+ \frac{L^3}{2EI} \left(\frac{P^2}{3} - P^2 \cos\theta + P^2 \cos^2\theta - PQ \sin\theta + Q^2 \sin^2\theta + 2PQ \cos\theta \sin\theta \right)$$
(1)

[3	mar	ks]
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Vertical deflection at position A, u_{v_A}

Differentiating (1) with respect to the applied load, *P*:

$$u_{\nu_{A}} = \frac{\partial U}{\partial P} = \frac{L^{3}}{6EI} (2P\cos^{2}\theta + Q\cos\theta\sin\theta) + \frac{L^{3}}{2EI} \left(\frac{2P}{3} - 2P\cos\theta + 2P\cos^{2}\theta - Q\sin\theta + 2Q\cos\theta\sin\theta\right)$$

[2 marks]

Setting dummy load to zero,

$$u_{\nu_{A}} = \frac{PL^{3}cos^{2}\theta}{3EI} + \frac{PL^{3}}{EI} \left(\frac{1}{3} - cos\theta + cos^{2}\theta\right)$$
$$= \frac{PL^{3}}{EI} \left(\frac{4}{3}cos^{2}\theta - cos\theta + \frac{1}{3}\right)$$
$$= \frac{16,000 \times 750^{3}}{225,000 \times 125,663.71} \left(\frac{4}{3}cos^{2}(55) - cos(55) + \frac{1}{3}\right)$$
$$\therefore u_{\nu_{A}} = 47.38 \text{ mm}$$



Horizontal deflection at position A, u_{h_A}

Differentiating (1) with respect to the dummy load, Q:

$$u_{h_{A}} = \frac{\partial U}{\partial Q} = \frac{L^{3}}{6EI} (2Qsin^{2}\theta + Pcos\thetasin\theta) + \frac{L^{3}}{2EI} (-Psin\theta + 2Qsin^{2}\theta + 2Pcos\thetasin\theta)$$

[2 marks]

Setting dummy load to zero,

$$u_{h_A} = \frac{L^3}{6EI} (P\cos\theta\sin\theta) + \frac{L^3}{2EI} (2P\cos\theta\sin\theta - P\sin\theta)$$
$$= \frac{PL^3}{EI} \left(\frac{7\cos\theta\sin\theta}{6} - \frac{1}{2}\sin\theta\right)$$
$$= \frac{16,000 \times 750^3}{225,000 \times 125,663.71} \left(\frac{7\cos(55)\sin(55)}{6} - \frac{1}{2}\sin(55)\right)$$

 $\therefore u_{h_A} = 33.09 \text{ mm}$